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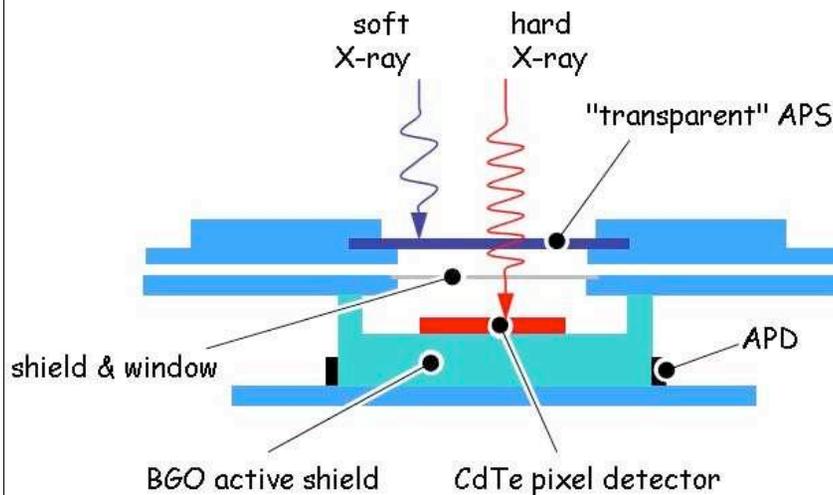
# Hard X-ray Imager (HXI) for the IXO mission

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# Onboard Configuration of WFI+HXI

“High energy extension for the non-thermal regime science”



## Requirements

Efficiency : ~100% up to 40 keV

Energy resolution : < 1 keV (FWHM)

FOV: similar to WFI (~14')

Position resolution : 1-3 arcsec (for 10-30" HPD)

Counting rate : 1 Crab /w <10% dead-time

Background :  $\sim 5 \times 10^{-4}$  c/s/keV/cm<sup>2</sup>

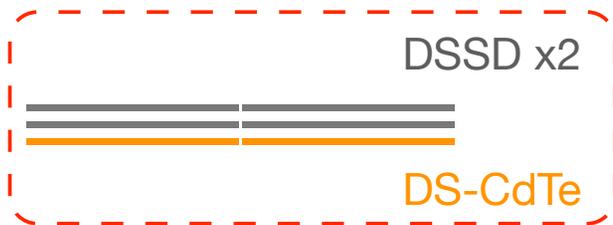
## Constraints for the I/F with WFI

Gap between the focal plane of WFI and surface of HXI should be smaller than 30 mm, which corresponds to the image defocus of ~10 arcsec.

Heat exchange onto WFI should be small (1-2W).

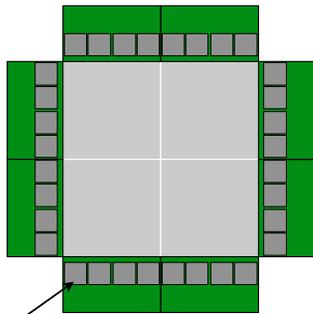
# HXI detector design

Based on the latest design of Hard X-ray Imager onboard ASTRO-H.



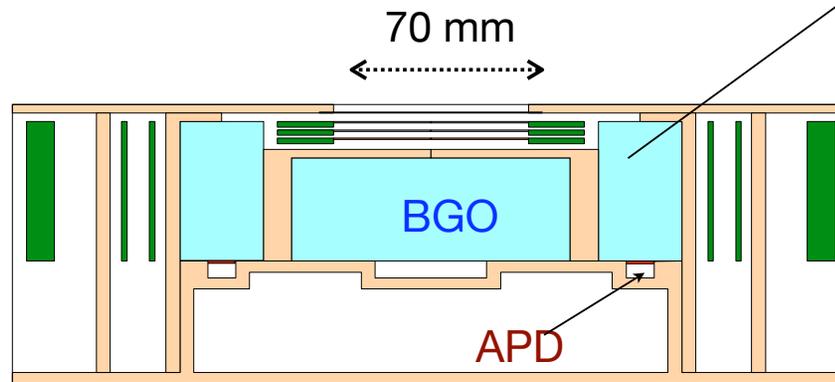
Main detection part consists of 2-layers of Double-Sided Silicon Strip (DSSD) and 1 layer of Double-sided CdTe strip (DS-CdTe). Each layer contains 2x2 devices.

Top-view  
(imager part)



Readout ASICs are directly coupled with devices within the same plate.

Cross-section-view



Active shield :  
2 cm thick BGO  
(default)



Heritage from  
Suzaku-HXD

Total Weight : 24 kg  
Total Power : ~32W

(including electronics)

# Current Design Parameters

Characteristics	Hard X-ray Imager
Detector Type	Si and CdTe Schottky Diode double sided strip
Strip pitch	220 um (for both side) (~2.3" @ FL20m)
Number of strips/channel	320 (for both side) total 640 strips, 1280 ch for CdTe only. With 2 DSSDs and a DS-CdTe, total of 1920 strips, 3840 ch.
Array Size (mm <sup>2</sup> )	70 × 70
Field of View	12 × 12 arcmin <sup>2</sup>
Energy range	10-80 keV
Energy Resolution	< 1 keV(FWHM)
Non X-Ray Background	5 × 10 <sup>-4</sup> counts/s/keV/cm <sup>2</sup> roughly flat (based on Suzaku)
Count rate/source with 10% pile-up	20k (counts/sec) for each strip
Timing accuracy	< 10 us
Typical/ Max telemetry	11 kbps / 1 Mbps (above 10 keV)
Operating Temperature	Detector -20 ± 5 °C (Minimum temperature -40 °C)
Total Mass	24 kg
Total Power	32 W (in operation)

(As of 2009-01-27)

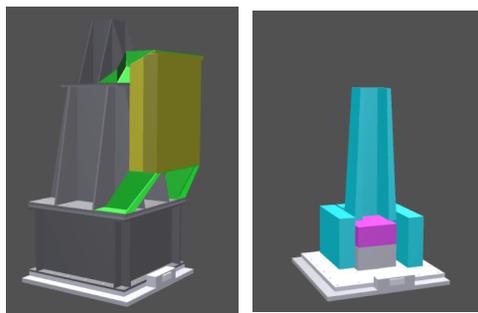
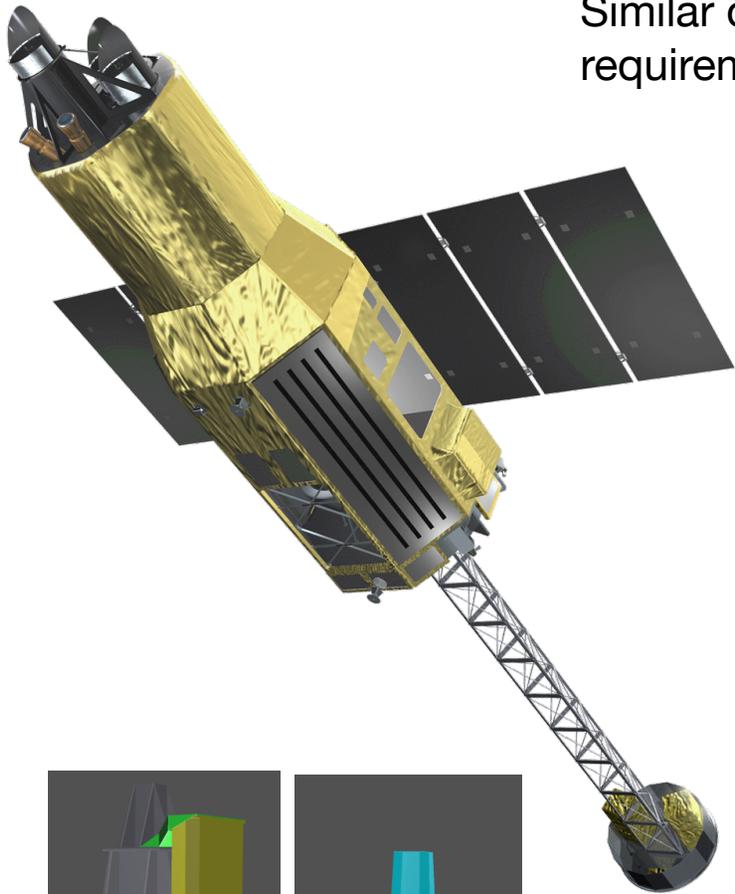


# HXI onboard ASTRO-H

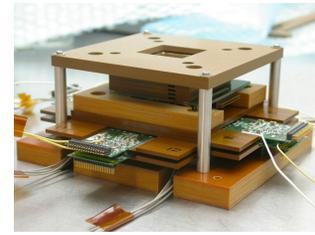
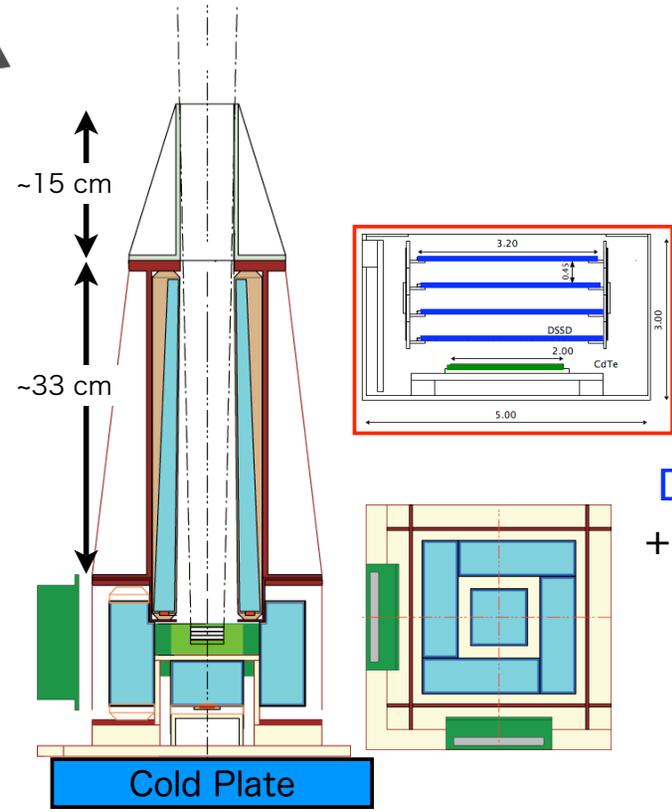


Similar design parameters derived from similar scientific requirements, except the following differences;

- 4 layers of DSSD + 1 layer of DS-CdTe
- Layer size 32x32 mm<sup>2</sup>
- Fully shielded with BGO scintillators (4cm thick)

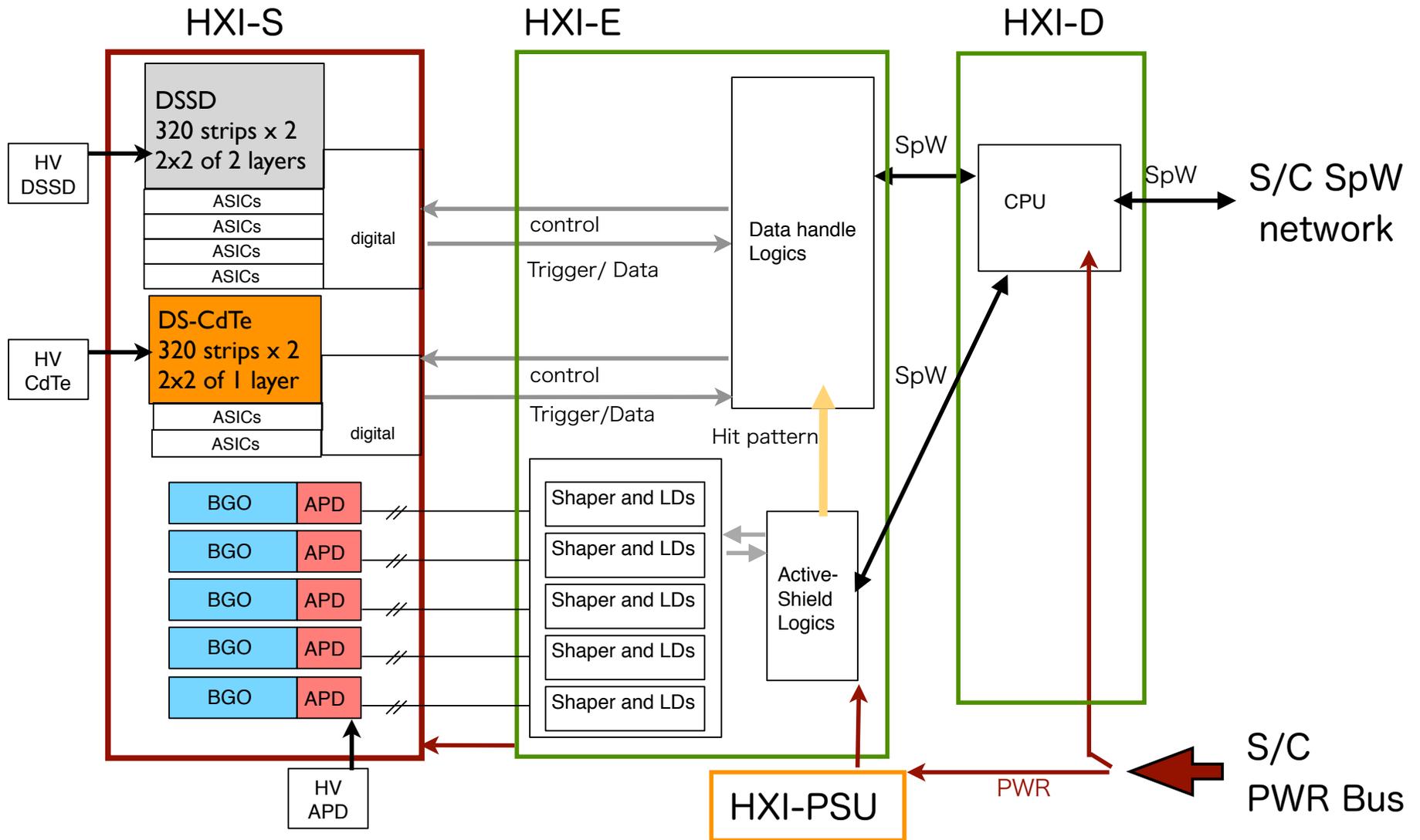


(c) MHI



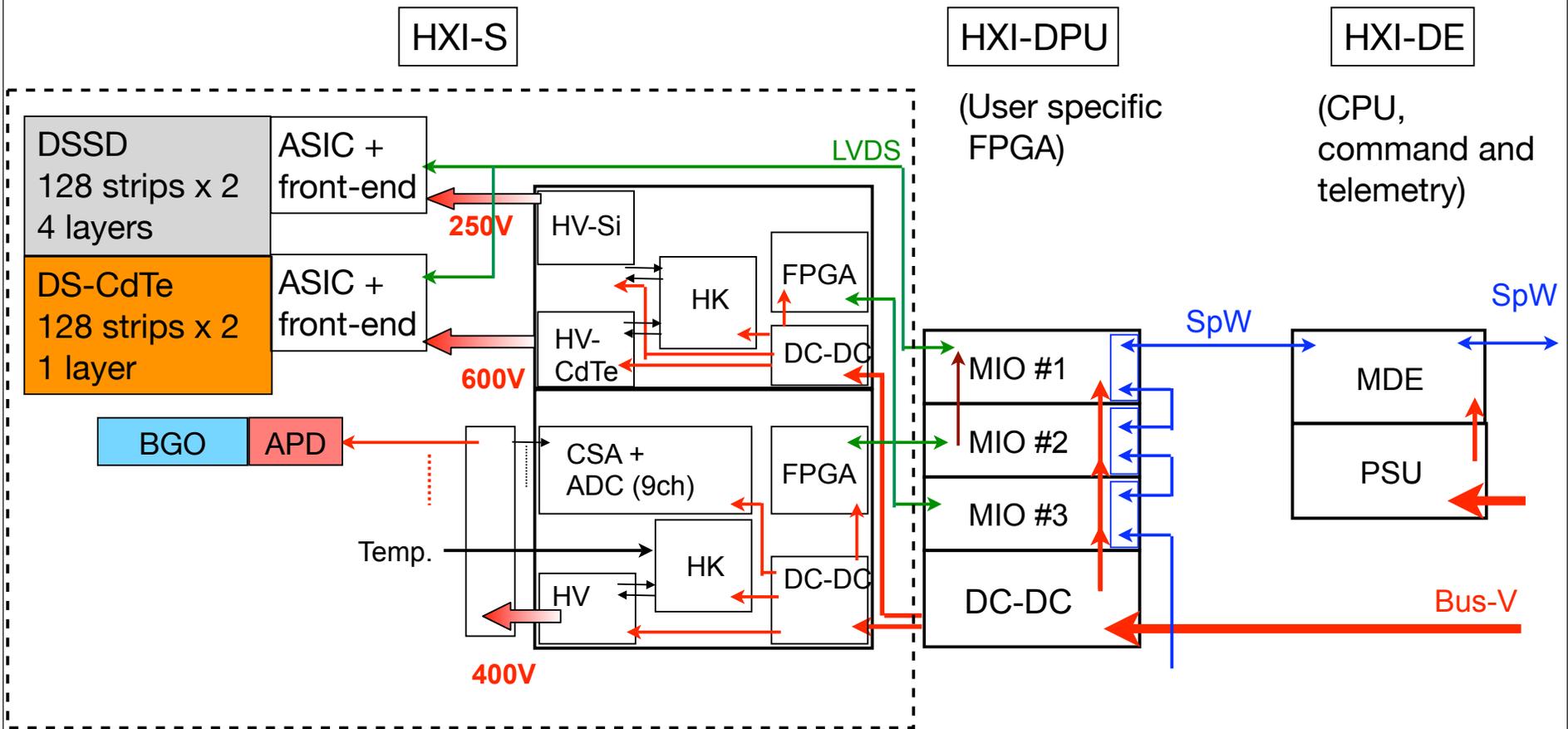
DSSD (0.5mm x 4)  
+ DS-CdTe (0.5mm)

# IXO-HXI block diagram (in PDD v5)





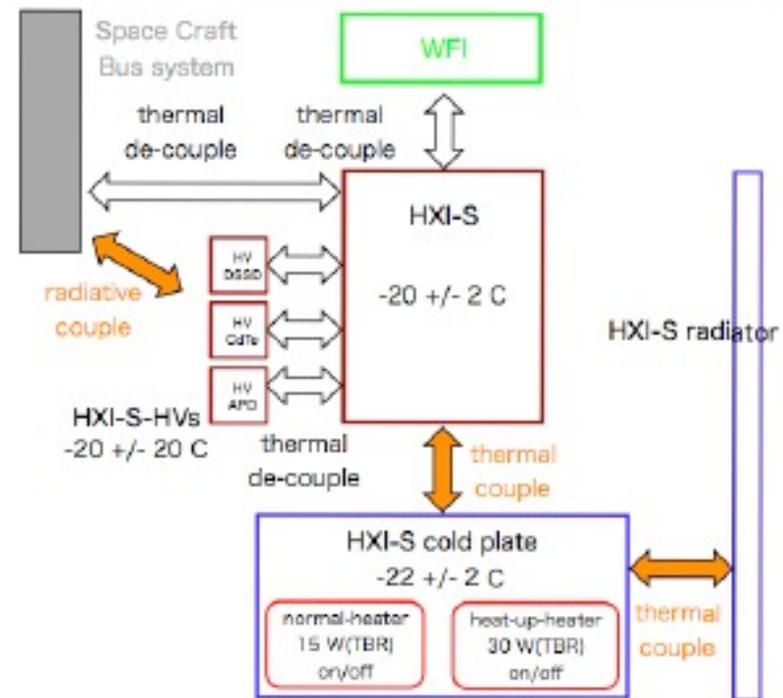
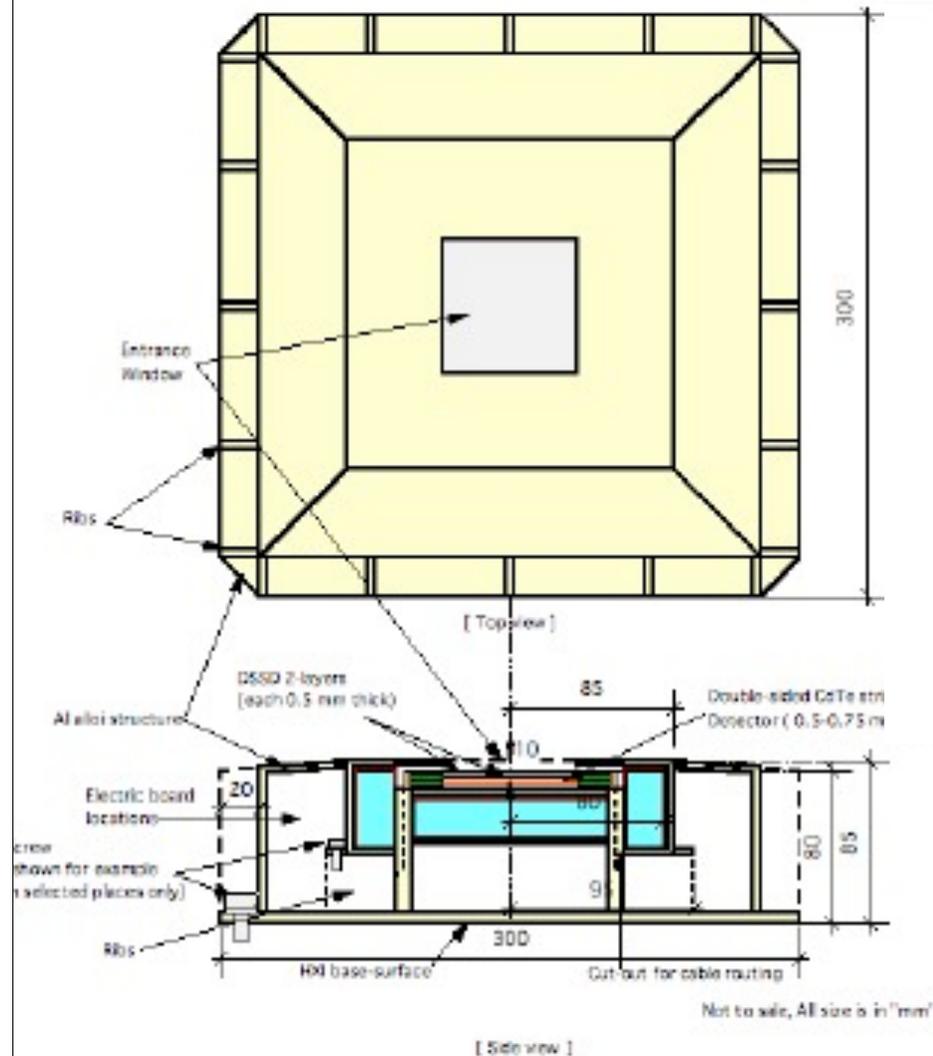
# Block Diagram of ASTRO-H/HXI



Analog part of electronics is "attached" as an extension box to HXI-S. Sensor readout is divided into two parts, main detection part (DSSD/CdTe) and shield part (APD). HK data like temperature and monitor voltage is also processed in this part.

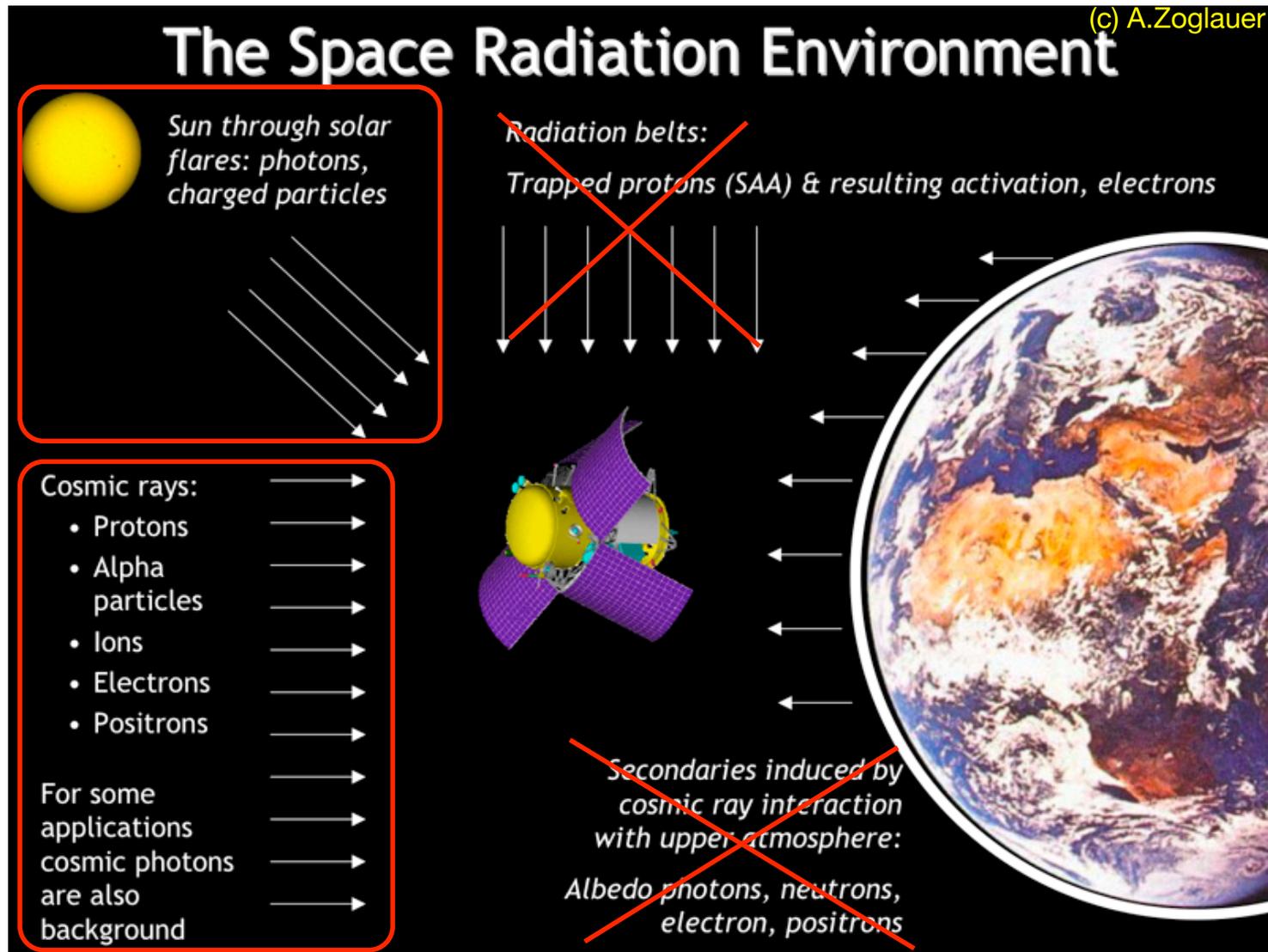
Digital processing part consists of user-customized FPGA part using a standard I/O boards and CPU board, communicating via SpW with each other.

# Mechanical/Thermal Design



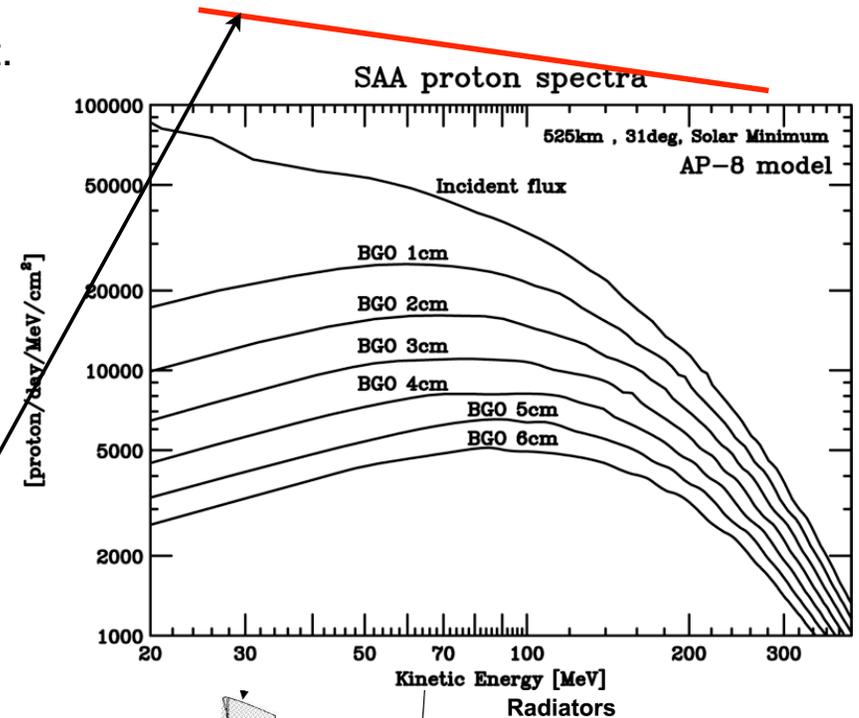
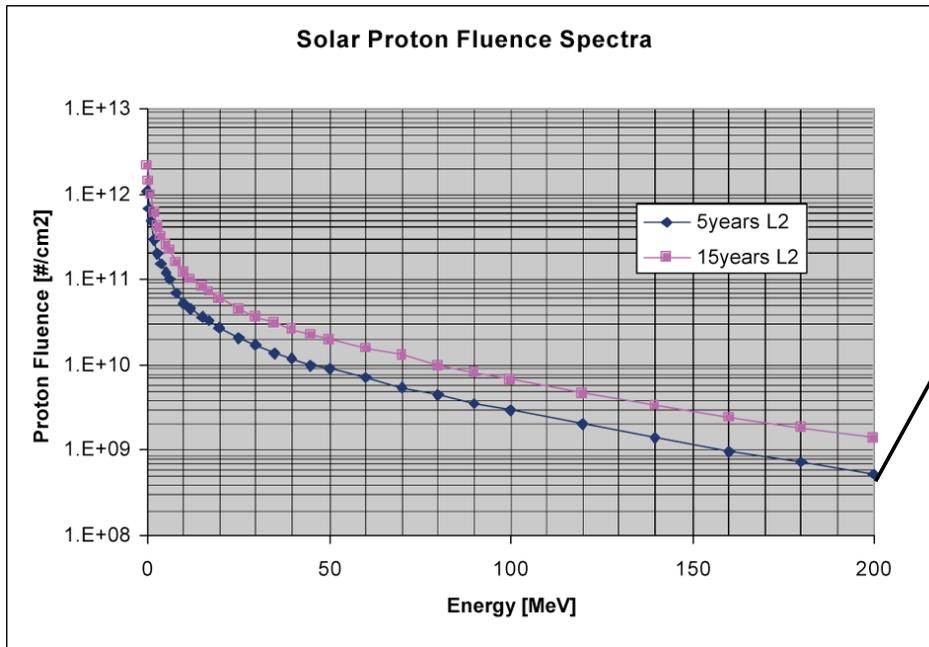
A housing structure made of CFRP will provide a sufficient mechanical strength. Temperature is controlled by a “cold plate” which is thermally coupled to the radiator and can be warmed with built-in heaters.

# Radiation Environment at L2



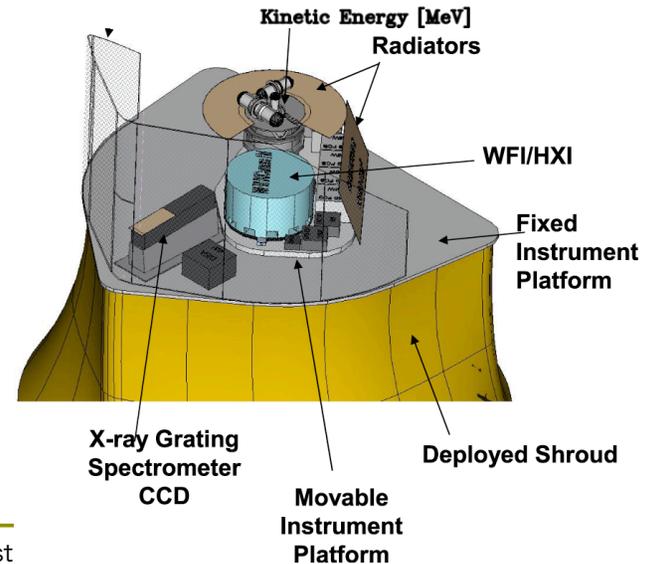
# Radiation Environment at L2

Solar Proton becomes the dominant component.



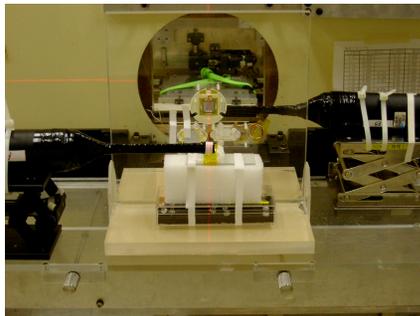
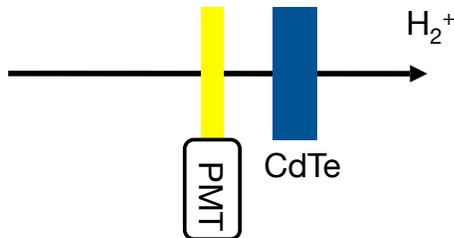
From "Xeus Environmental Specification" (Issue 1.0)

Since the coming direction of solar particles is known, a thick passive shield on the Sun direction would efficiently act as an absorber.



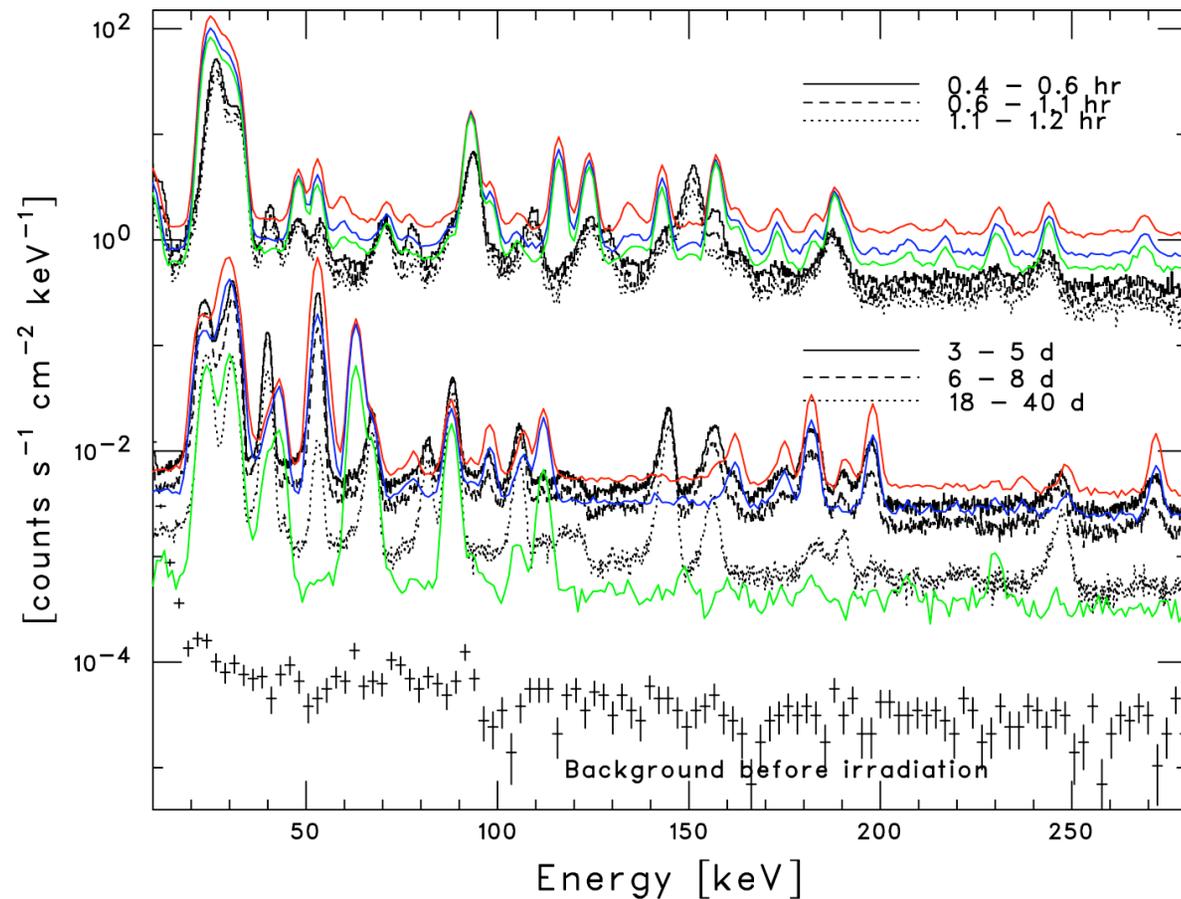
# Background Simulation

## Beam irradiation experiment

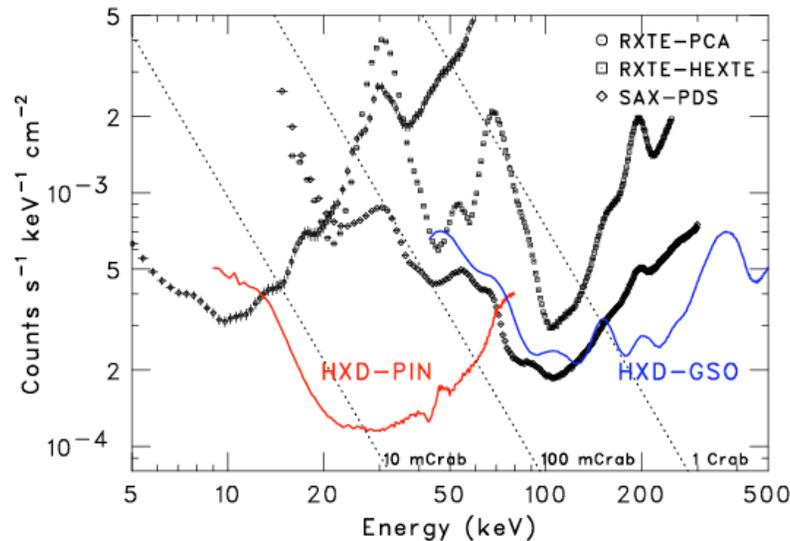


- Beam  
p150MeV (RCNP)  
1.7x10<sup>10</sup> proton / 87min
- Target :  
CdTe  
10x10x0.5 mm<sup>3</sup>
- Measurement :  
CdTe(ΔE~3keV)  
Cu+Pb cave

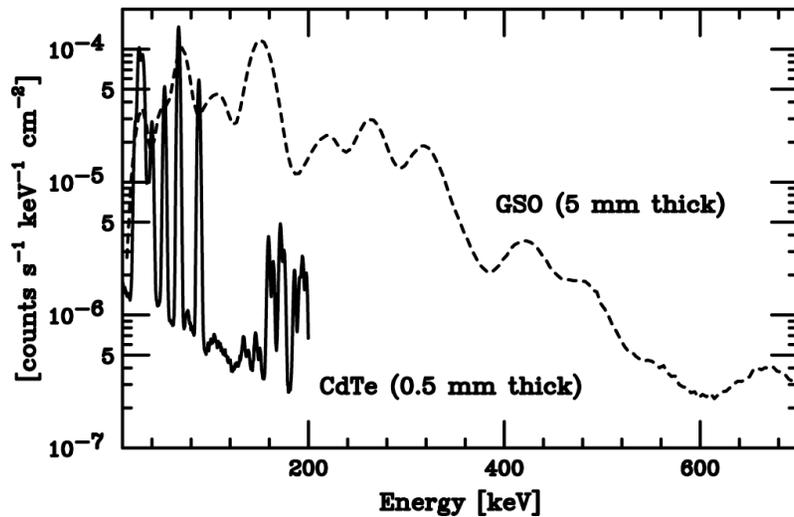
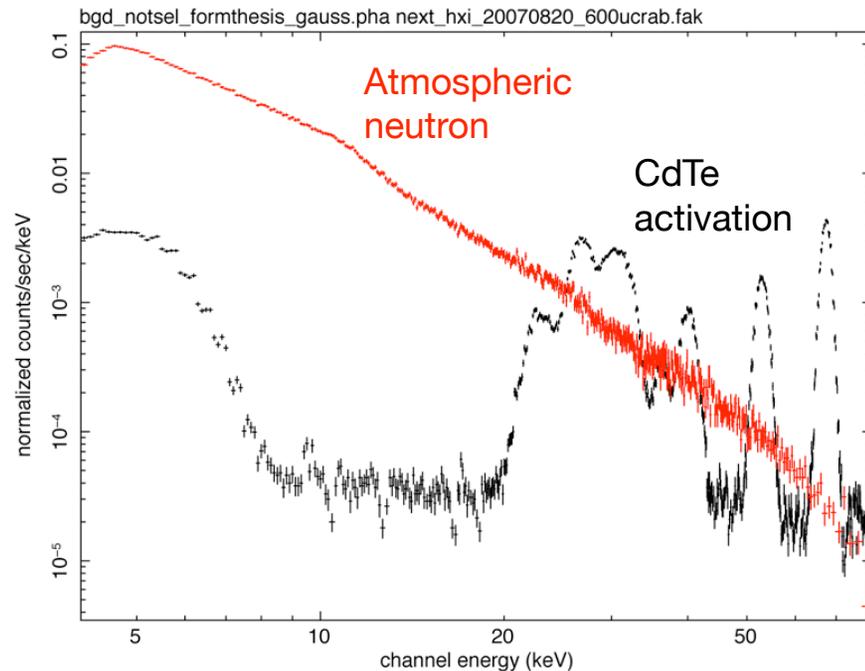
Comparison between the measurement and simulation of induced activations shows a reliable agreement except several peaks. We are now investigating the detail.



# Current Estimation



Estimated background spectrum with the MC simulation for ASTRO-H/HXI.

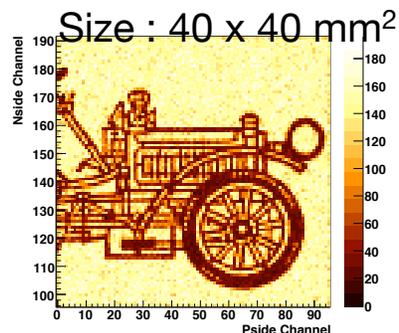
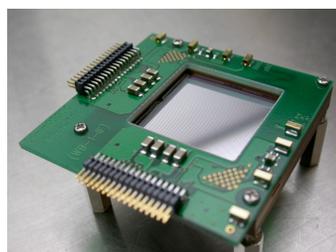


The neutron component will be negligible in case of L2 environment, but we need further detailed study with a mass model of S/C to take into account the secondary emissions.

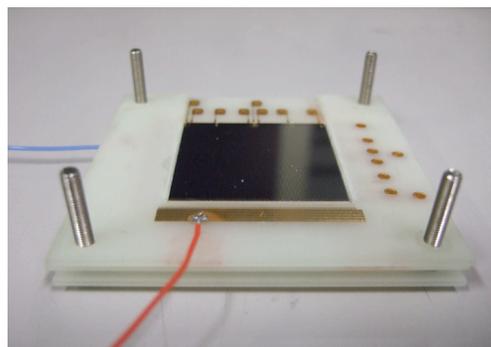
# Double-sided Si Strip Detector

Lower intrinsic background can be achieved than CdTe thanks to its small atomic number. Also act as a “fluorescent X-ray shield” for WFI located between CdTe and WFI. Background reduction with anti-coincidence can be also utilized.

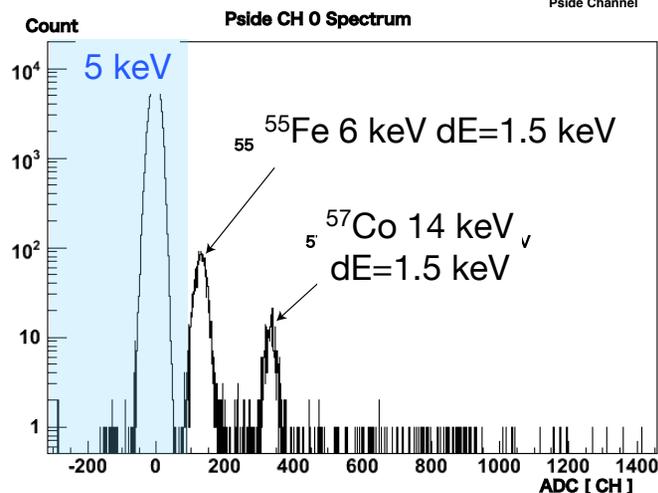
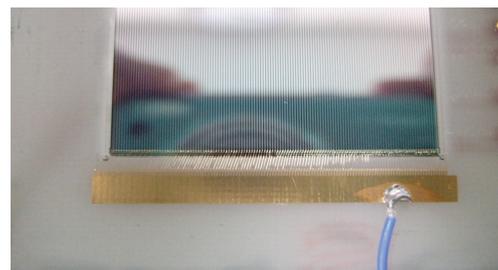
Engineering model Strip pitch: 400um



ASTRO-H fight model



Strip Pitch : 250μm  
Thickness: 500μm  
Area: 32 mm × 32 mm  
(128 ch for both sides)

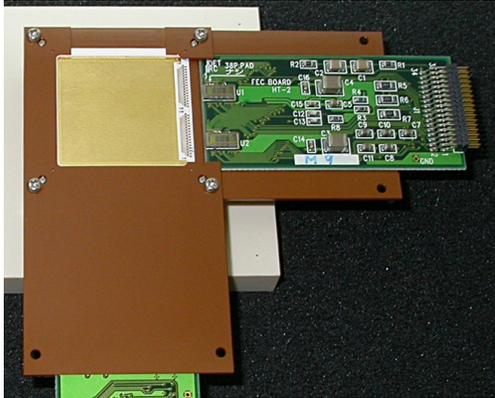


Low noise performance has been verified with a lower energy threshold of ~5 keV.

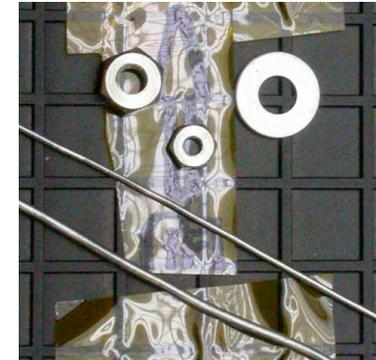
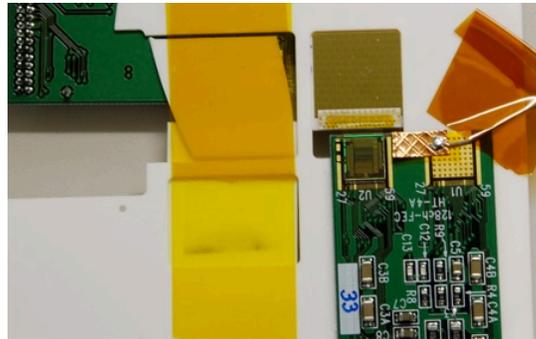
# Status of CdTe imager

Double-sided strip detector based on CdTe diode devices from ACRORAD

2.5 cm DS-CdTe



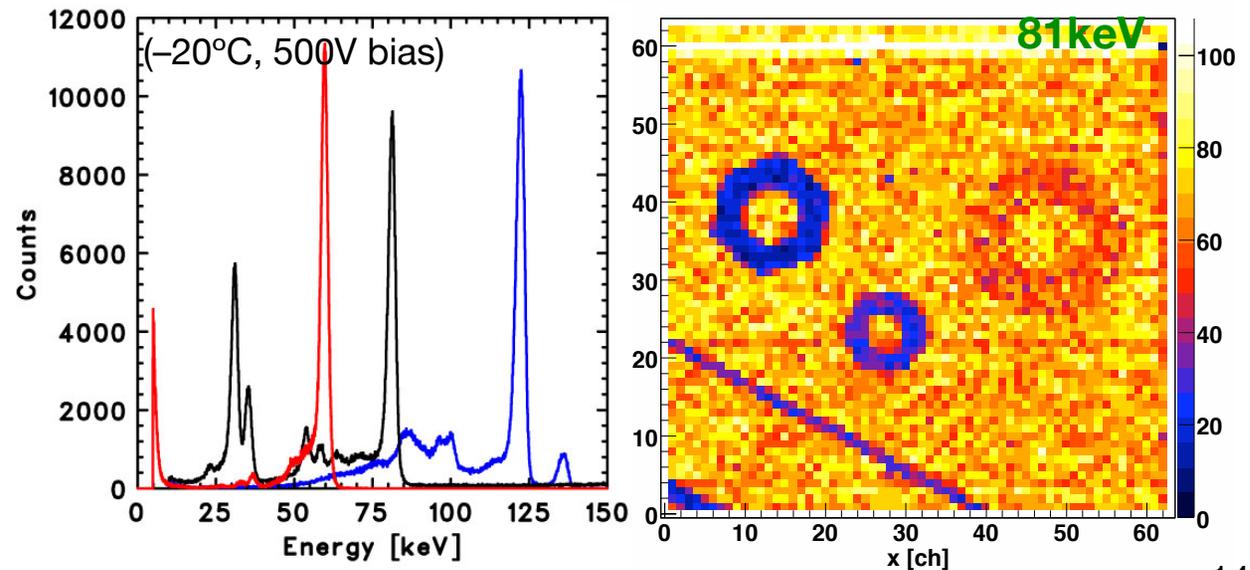
1.3 cm DS-CdTe



Shadow Image with various RIs ( $^{241}\text{Am}$ ,  $^{133}\text{Ba}$ ,  $^{57}\text{Co}$ )

## Comparisons of strip with Pixel

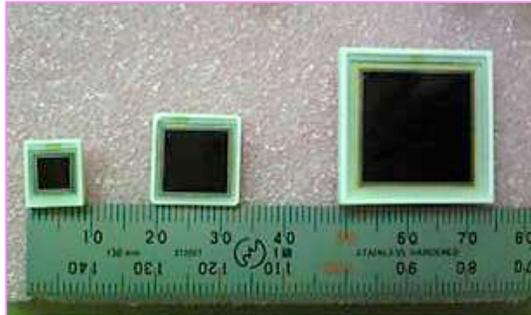
- smaller read-out channels  
( $2n$  instead of  $n^2$ )
- 1-dimensional ASIC
- relatively large leak and C



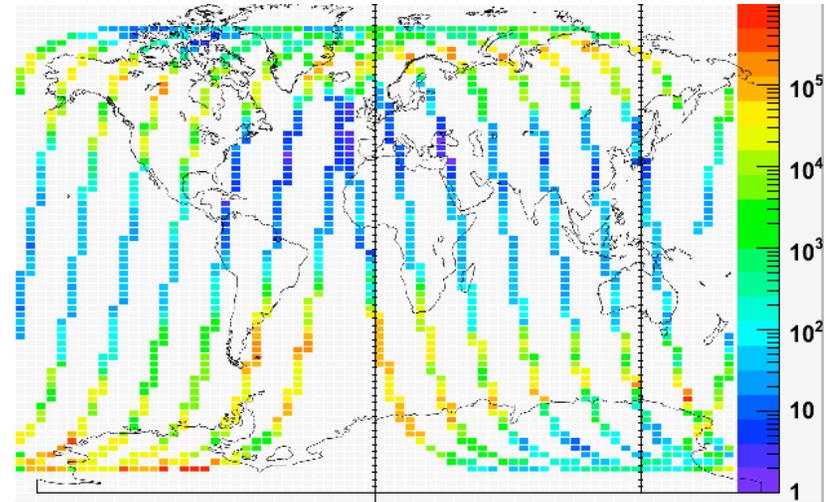
# APD for BGO read-out

## Reverse-type APD development (with HPK)

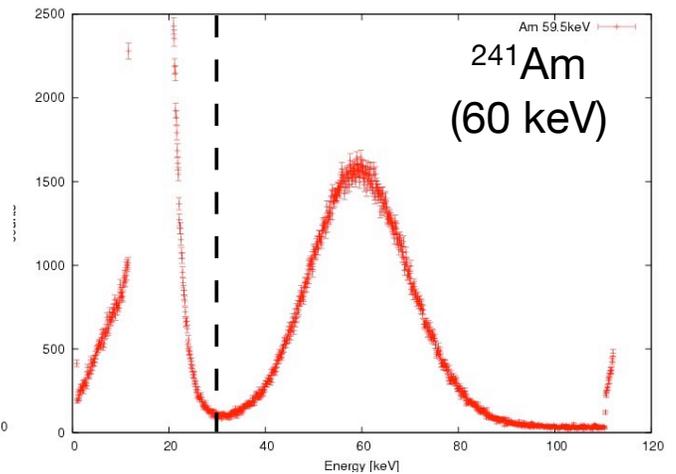
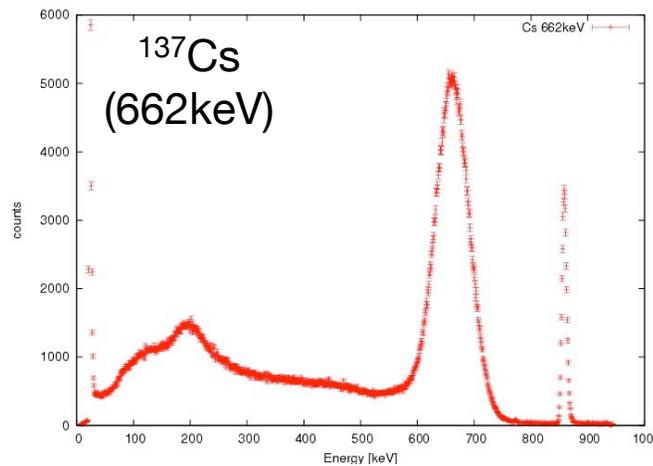
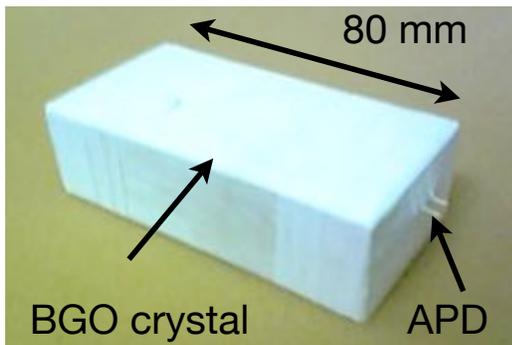
Ikagawa, et al. 2003, 2005 NIM-A  
Nakamoto et al. 2005 NIM-A,  
Sato R. et al., 2005 NIM-A,  
Kotoku et al. 2005 SPIE, 2006 NIM-A,  
Kataoka et al. 2006 NIM-A etc...



## APD demonstration in-orbit (Cute-1.7)



## Direct attachment



Energy threshold of ~30 keV will be achieved.

# Electronics with SpaceWire

SpaceWire based readout system are almost established with BBM for ASTRO-H.

Universal I/O board  
with SpW I/F



Sensor(DSSD/CdTe) and  
read-out system based on  
the SpW, developed for the  
balloon experiment



SpaceCube-I

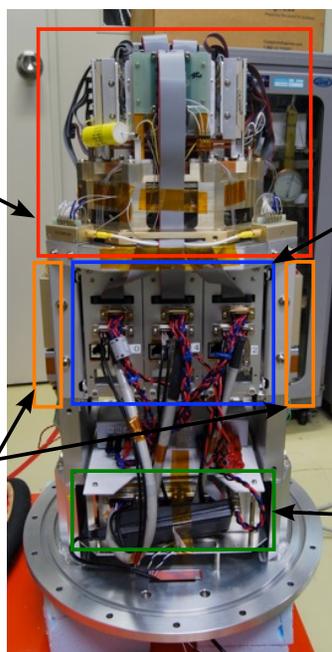


Main Detector  
Part

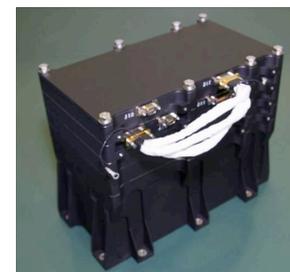
DSSD/CdTe  
Readout Part

H.V. Supply  
Part

SpaceWire  
Control Part



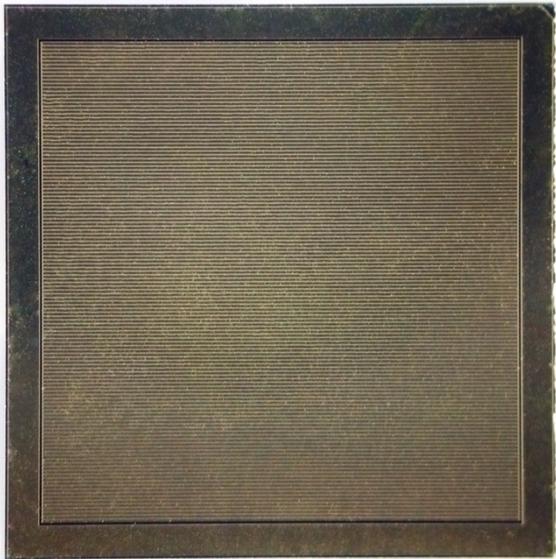
Technology demonstration satellite SDS-I  
of JAXA, launched as a piggy-back of “Ibusu  
(GOSAT)”, 4 days ago. A SpaceWire  
experimental module is onboard.



# Future Options

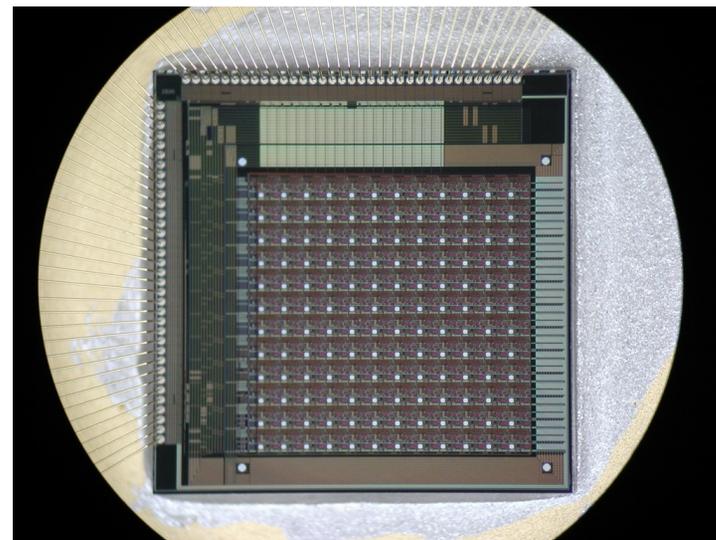
In the time-frame of IXO launch (2020's), we can expect further advanced technologies will be available.

Two examples are shown below. Also refer to Olivier's talk.



Fine pitch double-sided strip (60  $\mu\text{m}$ )

We are trying to make the position resolution finer than 100  $\mu\text{m}$ .  
Though a smaller HPD is required for HXT, 60  $\mu\text{m}$  pitch corresponds to an angular resolution of sub-arcsec.



2D low-noise ASIC (H04)

Next generation analog ASICs (H04; 12 x 12 channels with 270  $\mu\text{m}$  pitch) developed in JAXA (G.Sato et al. IEEE 2008).  
A low noise performance of 50 e<sup>-</sup> (@ 0 pF) is achieved.

# Another possibility

## CALISTE 64 DESIGN

- A hybrid component based on a 3D Plus space proof technology.

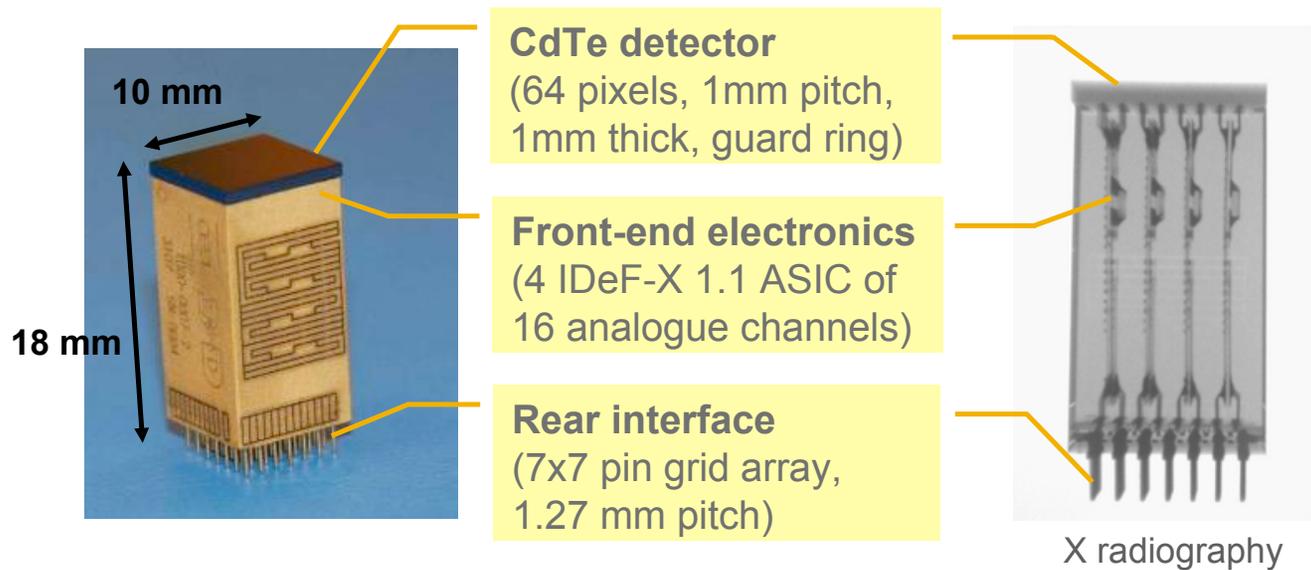
I r f u

cea

saclay

cnes

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plus

- Four micro PCB perpendicular to detection plan.
- Four ASICs to read out 2 rows of eight pixels each.
- Lateral routing to share signals between ASICs.

# Summary

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Hard X-ray imaging option will realize a large effective area with a fine resolution (10”), will provide a crucial capability to the IXO mission.

As for the HXI detector for IXO, the ASTRO-H/HXI based detector can be a default with a moderate TRL.

There are several things to be studied with more detail;

- Detailed interface definition (mech., elec., thermal)
- Detailed mechanical designing
- Calibration and Alignment plan (on ground and in orbit)
- Ground test requirements (EMC and verification @ RT etc ...)
- WFI-HXI interference (thermal, mechanical, shielding etc.. )
- BGD estimation with L2 environment (Geant-4 based)